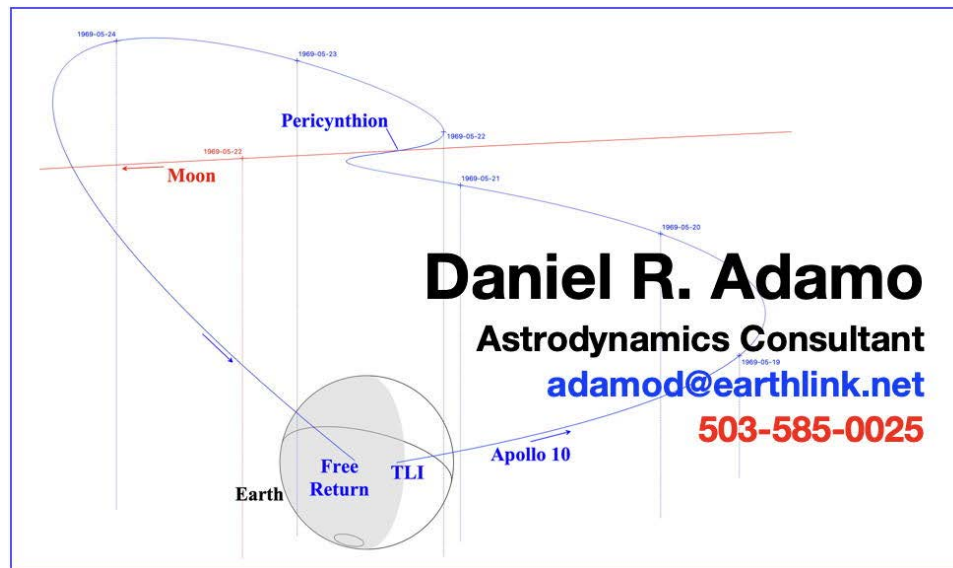


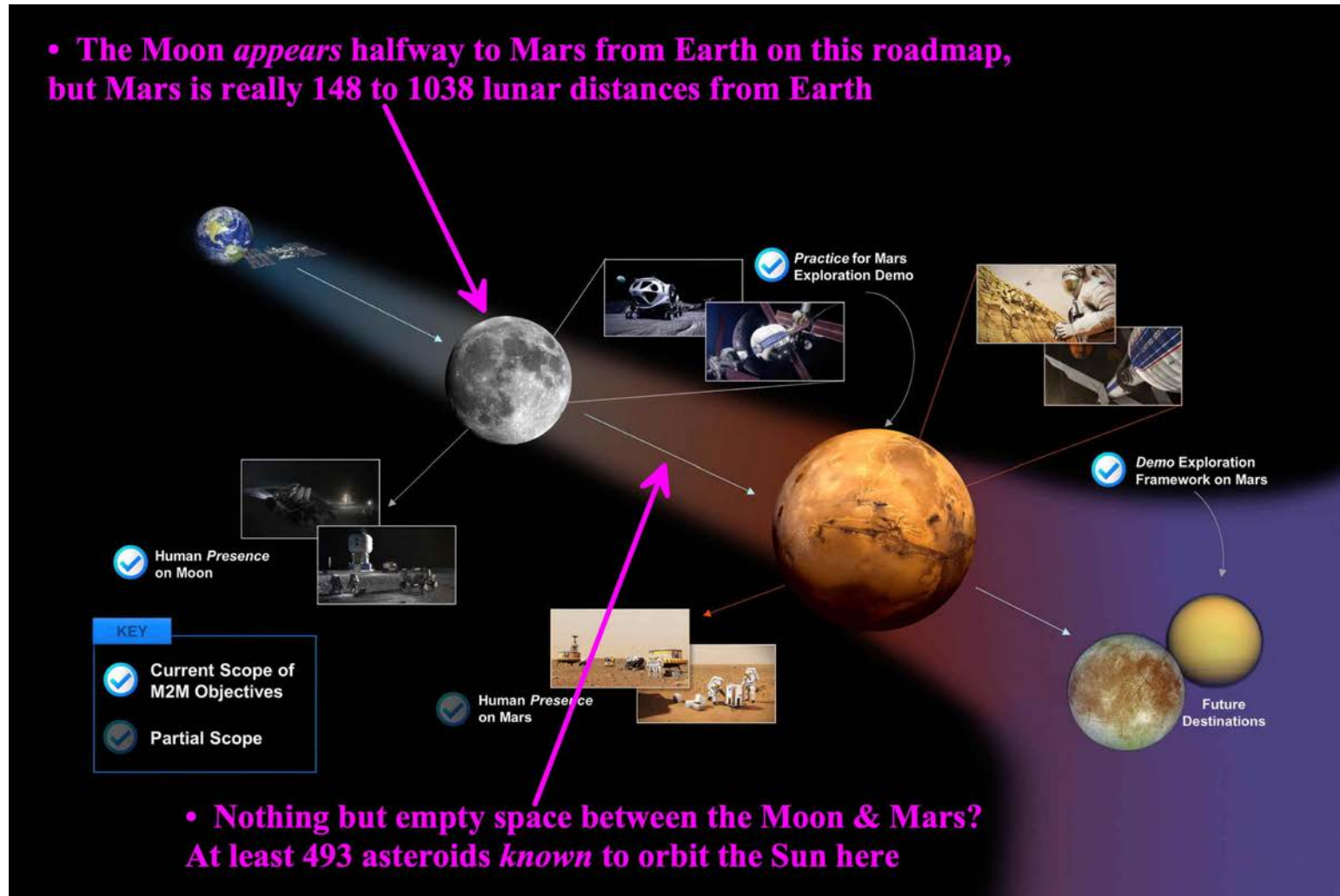
*Spanning The Earth/Mars Chasm  
With Interplanetary Synodic-Resonant Waypoints*



**AIAA Houston Section Lunch 'n Learn (LnL)**  
18 March 2025

# Interplanetary Synodic-Resonant Waypoints (WPs)

## The Chasm: What's Wrong With NASA's Moon to Mars (M2M) Roadmap?<sup>1</sup>



<sup>1</sup> Reference [https://www.nasa.gov/wp-content/uploads/2023/04/m2m\\_strategy\\_and\\_objectives\\_development.pdf?emrc=c21aff?emrc=c21aff](https://www.nasa.gov/wp-content/uploads/2023/04/m2m_strategy_and_objectives_development.pdf?emrc=c21aff?emrc=c21aff), p. 6 (accessed 12 Mar 2025).

# Interplanetary Synodic-Resonant Waypoints (WPs)

## No Wonder We've Remained About 20 Years From Landing Humans On Mars Since Dr. Wernher von Braun First Addressed The Problem in 1948<sup>2</sup>

- Mars is orders of magnitude more remote than the Moon: add a "not to scale" disclaimer to the M2M roadmap and reduce mistaken "Mars is easy" expectations among stakeholders
- Add interplanetary "waypoints" (WPs) to the M2M roadmap and reduce travel far from aid/haven

## Transport Technology Solves Difficult Problems By Decomposing Them Into More Tractable "Stages"

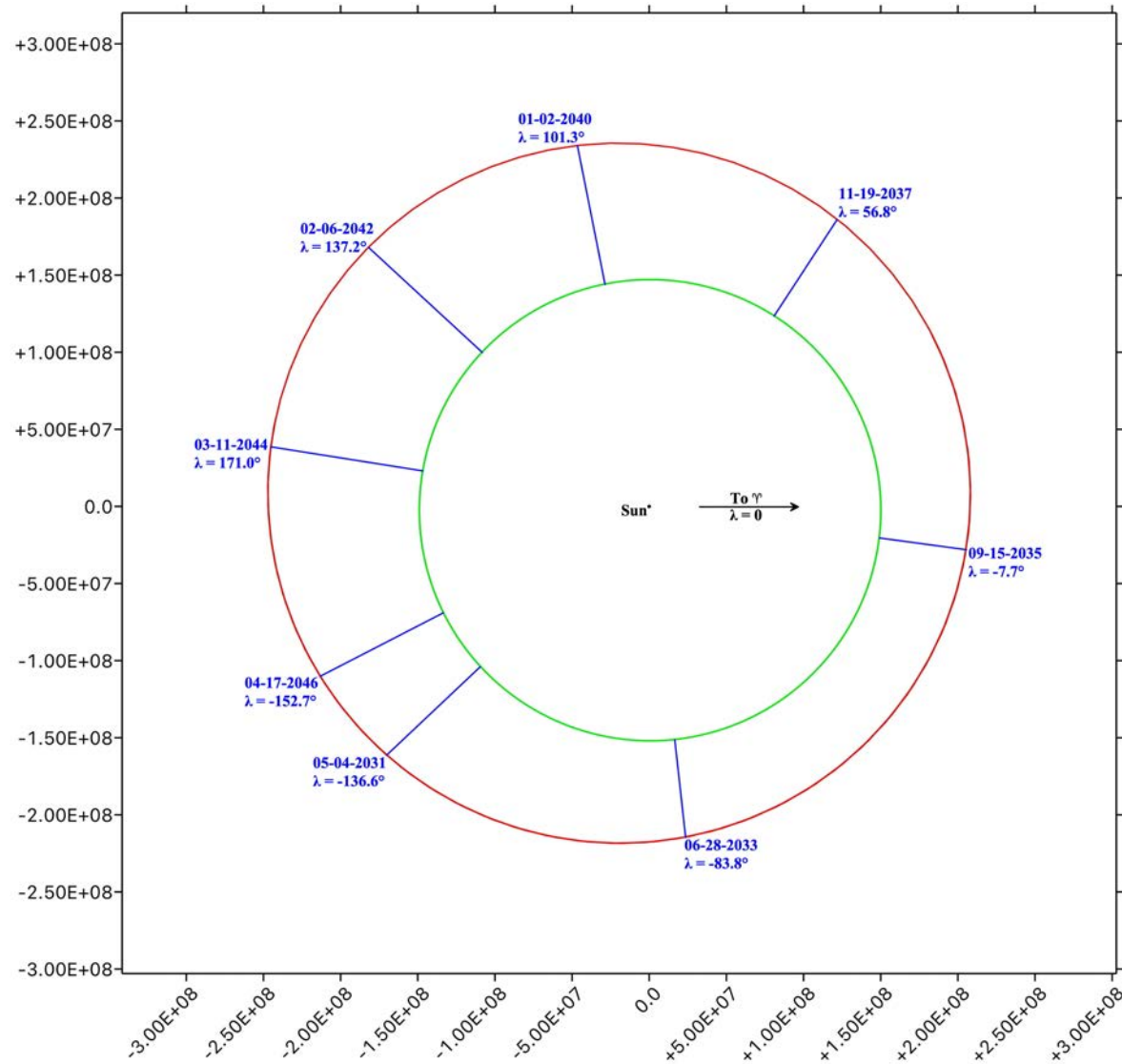
- Before railroads, *stagecoaches* swapped spent horses for fresh ones at strategic depots
- Missouri-to-California mail delivery via Pony Express relay in about 10 days circa 1860
- Single stage to Earth orbit capability remains elusive: only multi-stage rockets in use today
- If it's so difficult/dangerous, why do we insist on flying humans from Earth to Mars nonstop?
  - Place WPs in orbit about the Sun between Earth and Mars at which to routinely loiter en route, topping off consumables and optionally delivering or supplementing payload mass
  - Select WP orbits providing maximum availability naturally, with minimal propulsion

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<sup>2</sup> Reference [https://en.wikipedia.org/wiki/The\\_Mars\\_Project](https://en.wikipedia.org/wiki/The_Mars_Project) (accessed 12 Mar 2025).

# Interplanetary Synodic-Resonant Waypoints (WPs)

## Earth/Mars Transit Opportunities Arise When Mars Is Near Opposition

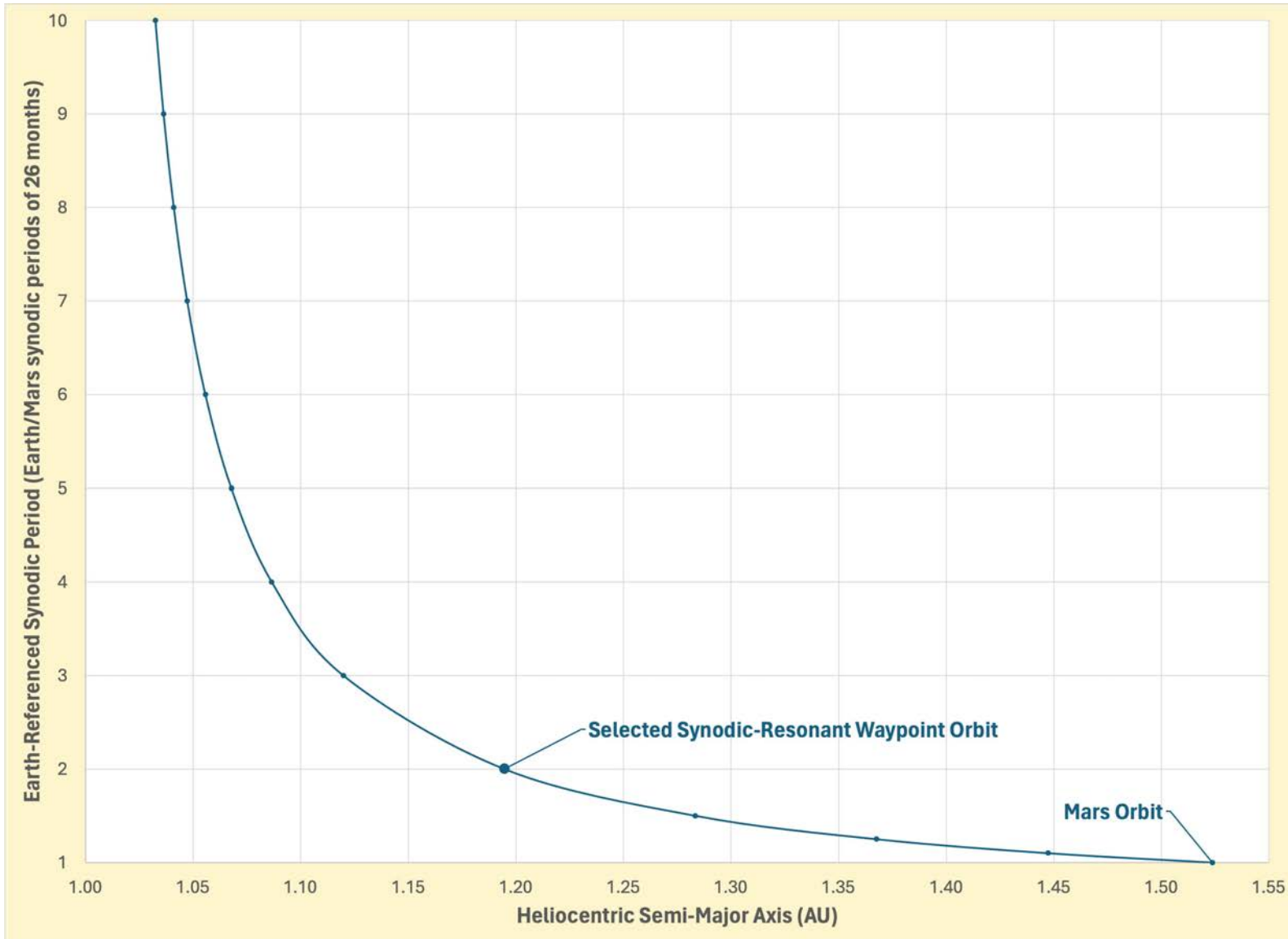


km Units J2KE Coordinate System  
 Origin: Sun (10) Red Plot: Mars (499) Green Plot: Earth (399)  
 Plot-To-Screen Transformation: Yaw = +0.00° Pitch = +0.00° Roll = +0.00°

- Average time between oppositions is 26 months, the synodic period  $\tau$  of Mars referenced to Earth
- Because  $\tau$  is 2 months more than 2 years, successive oppositions advance in ecliptic longitude  $\lambda$  by about  $360 \cdot 2/12 = 60^\circ$
- Due to Mars orbit eccentricity, intervals between opposition dates vary as they cycle through  $\lambda$  over 15 years, a *synodic sequence*  $T_s$

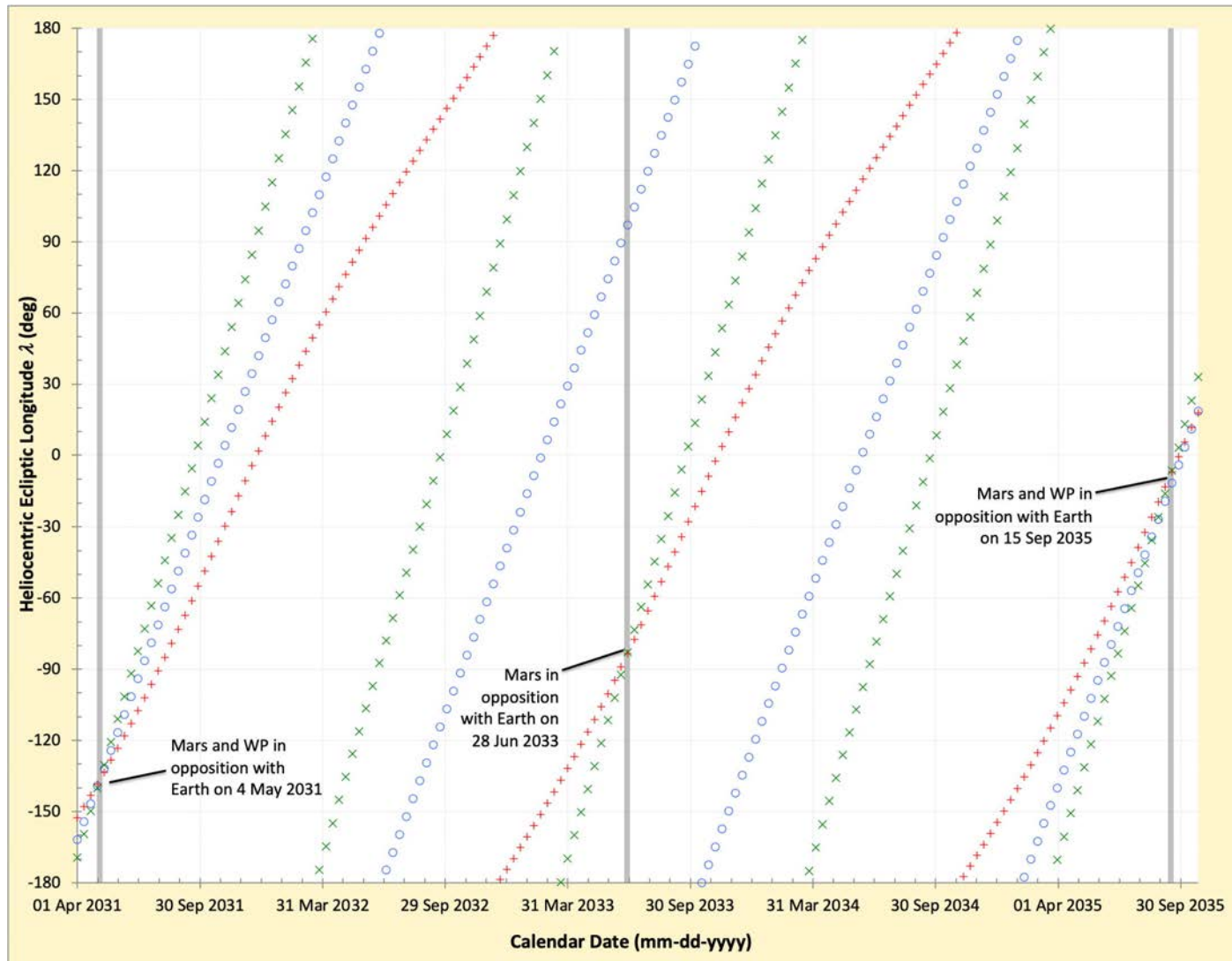
# Interplanetary Synodic-Resonant Waypoints (WPs)

The Selected WP Orbit Is Available During 50% Of Earth/Mars Oppositions



# Interplanetary Synodic-Resonant Waypoints (WPs)

A WP Availability Cycle Plays Out Over  $2\tau = 52$  Months = 4.27 Years



- X = Earth
  - + = Mars
  - O = WP
- with  $\lambda$  at  
 05-04-2031  
 opposition  
 $\lambda_{31} = -136.6^\circ$

# Interplanetary Synodic-Resonant Waypoints (WPs)

## In General, A $\lambda_{31} = -136.6^\circ$ WP Is Suboptimal With Respect To Earth/Mars Transits

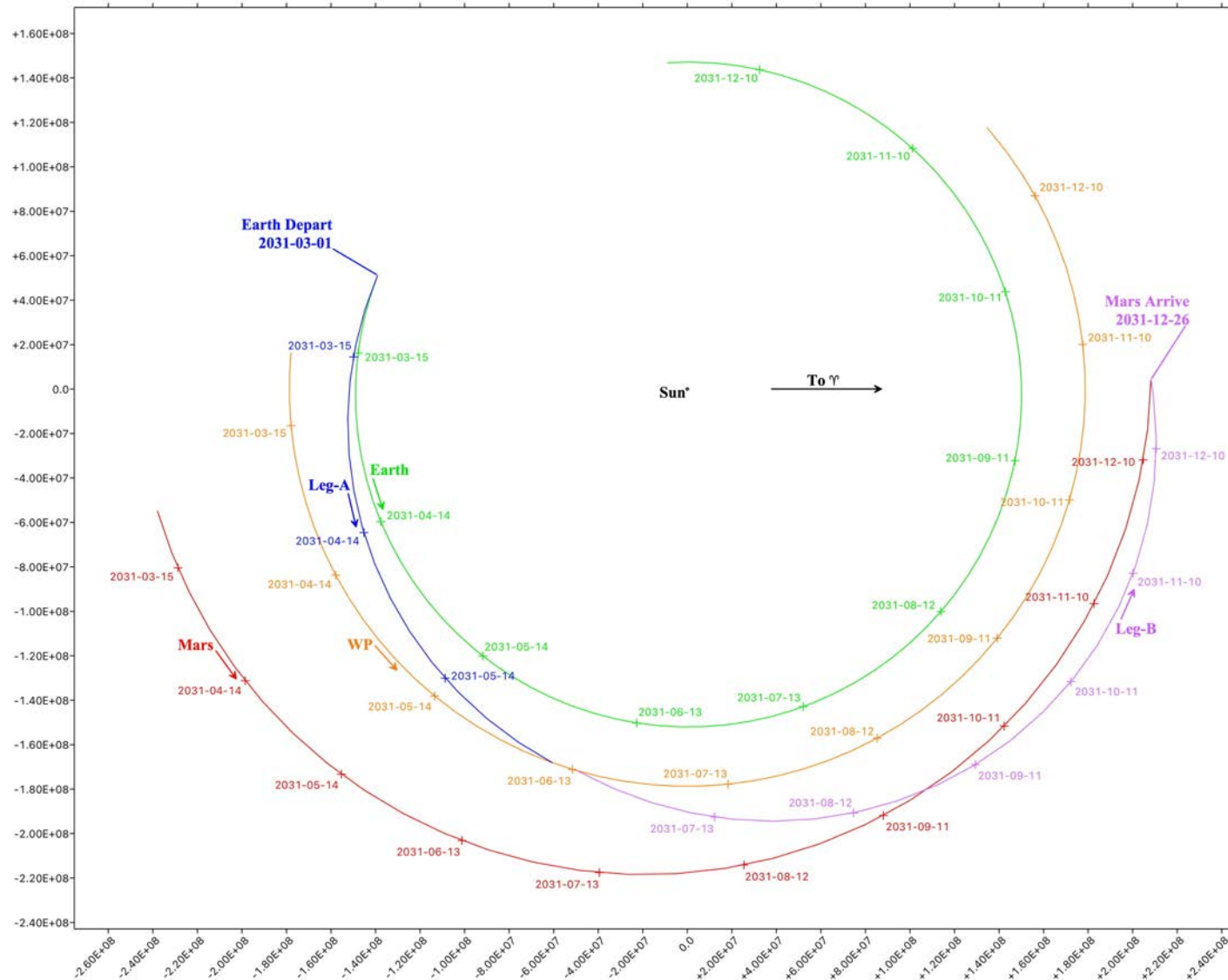
- Define an optimality metric  $\sum v_\infty$  as the sum of four hyperbolic asymptote speeds
  - Earth-to-Mars transits are composed of Leg-A, a WP loiter, and Leg-B stages, while Mars-to-Earth transits are composed of Leg-C, a WP loiter, and Leg-D stages
  - Earth-to-Mars (Leg-AB) transit  $\sum v_\infty = v_{\infty Adep} + v_{\infty Arr} + v_{\infty Bdep} + v_{\infty Barr}$
  - Mars-to-Earth (Leg-CD) transit  $\sum v_\infty = v_{\infty Cdep} + v_{\infty Carr} + v_{\infty Ddep} + v_{\infty Darr}$
- Use the method of embedded trajectory grids to evaluate  $\sum v_\infty$  for thousands of Leg-AB or Leg-CD transits using 2-Leg Lambert (2LL) scans at 5-day granularity
  - Comparable to NHATS methodology<sup>3</sup>, but a roundtrip itinerary starting at Earth is not imposed
  - Compliancy limits include maximum time-of-flight, minimum WP loiter, and maximum  $\sum v_\infty$
  - Step through an opposition season, computing Leg-A/C transits for various WP arrival dates
  - For each compliant Leg-A/C transit, compute compliant Leg-B/D trajectories for various WP loiter times and arrival dates
  - Identify the compliant Leg-AB/CD transit with minimum  $\sum v_\infty$  during the 2LL scan

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<sup>3</sup> Reference <https://cneos.jpl.nasa.gov/nhats/intro.html> (accessed 12 Mar 2025).

# Interplanetary Synodic-Resonant Waypoints (WPs)

## The $\lambda_{31} = -136.6^\circ$ WP Is Over-Phased For Earth-To-Mars Transits

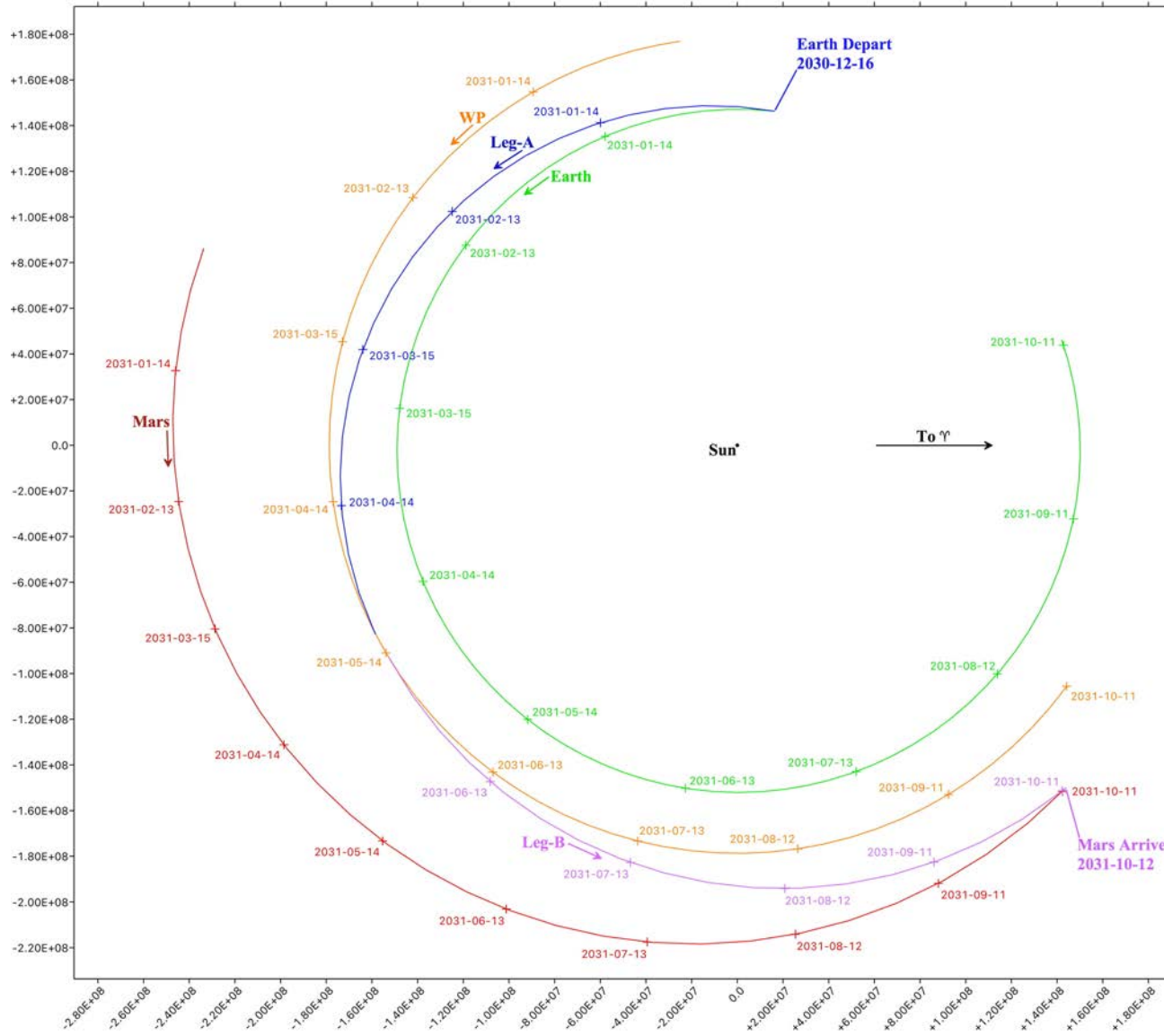


- Plot is of minimum  $\sum v_\infty$  Leg-AB transit circa 2031 for this WP
- Leg-B must cruise outside the orbit of Mars for more than 90 days because Leg-A had to leave Earth too late



# Interplanetary Synodic-Resonant Waypoints (WPs)

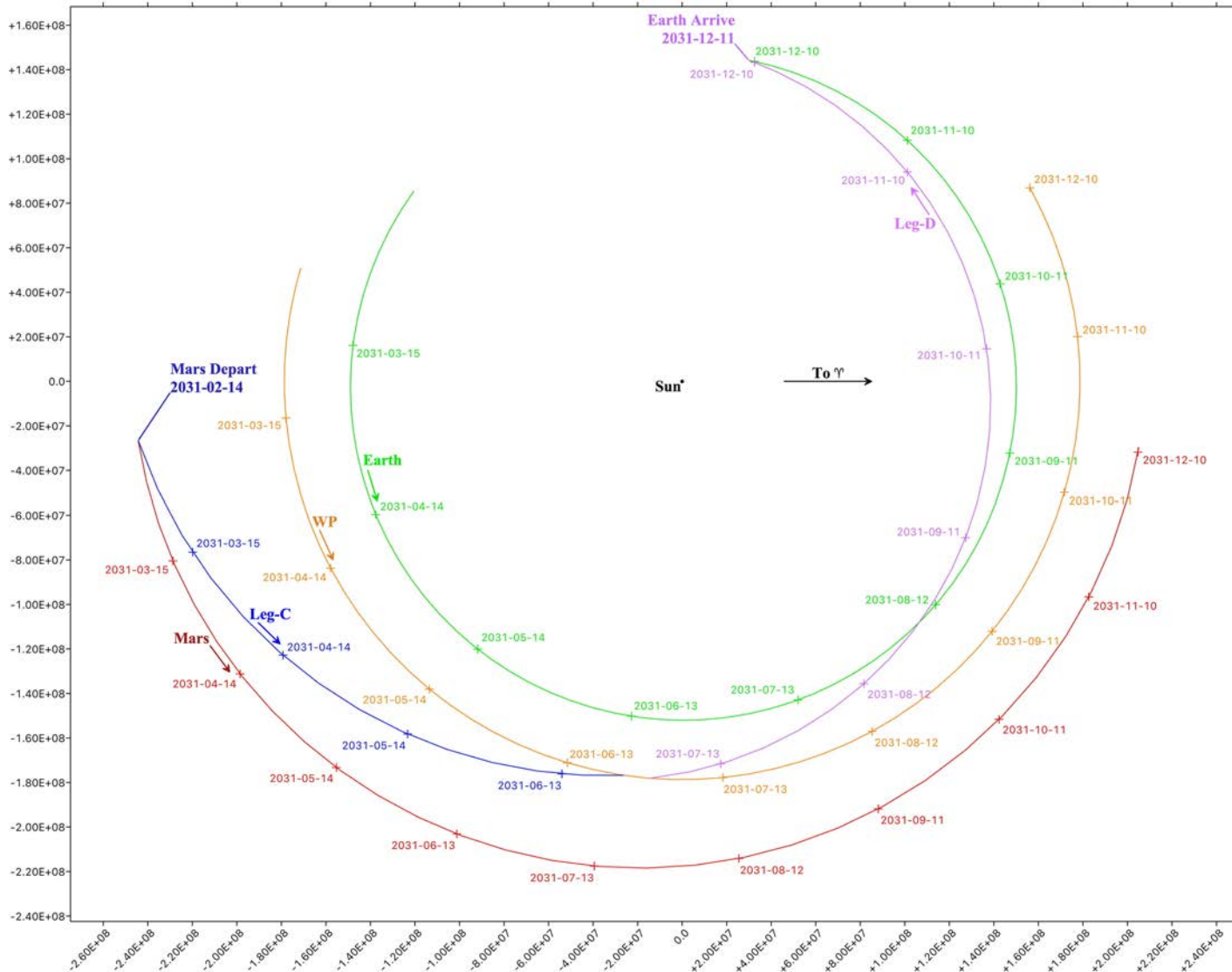
## The $\lambda_{31} = -156.6^\circ$ WP Produces Most Efficient Earth-To-Mars Transits



- Plot is of minimum  $\sum v_\infty$  Leg-AB transit circa 2031 for this WP
- Note this transit always remains between the orbits of Earth and Mars

# Interplanetary Synodic-Resonant Waypoints (WPs)

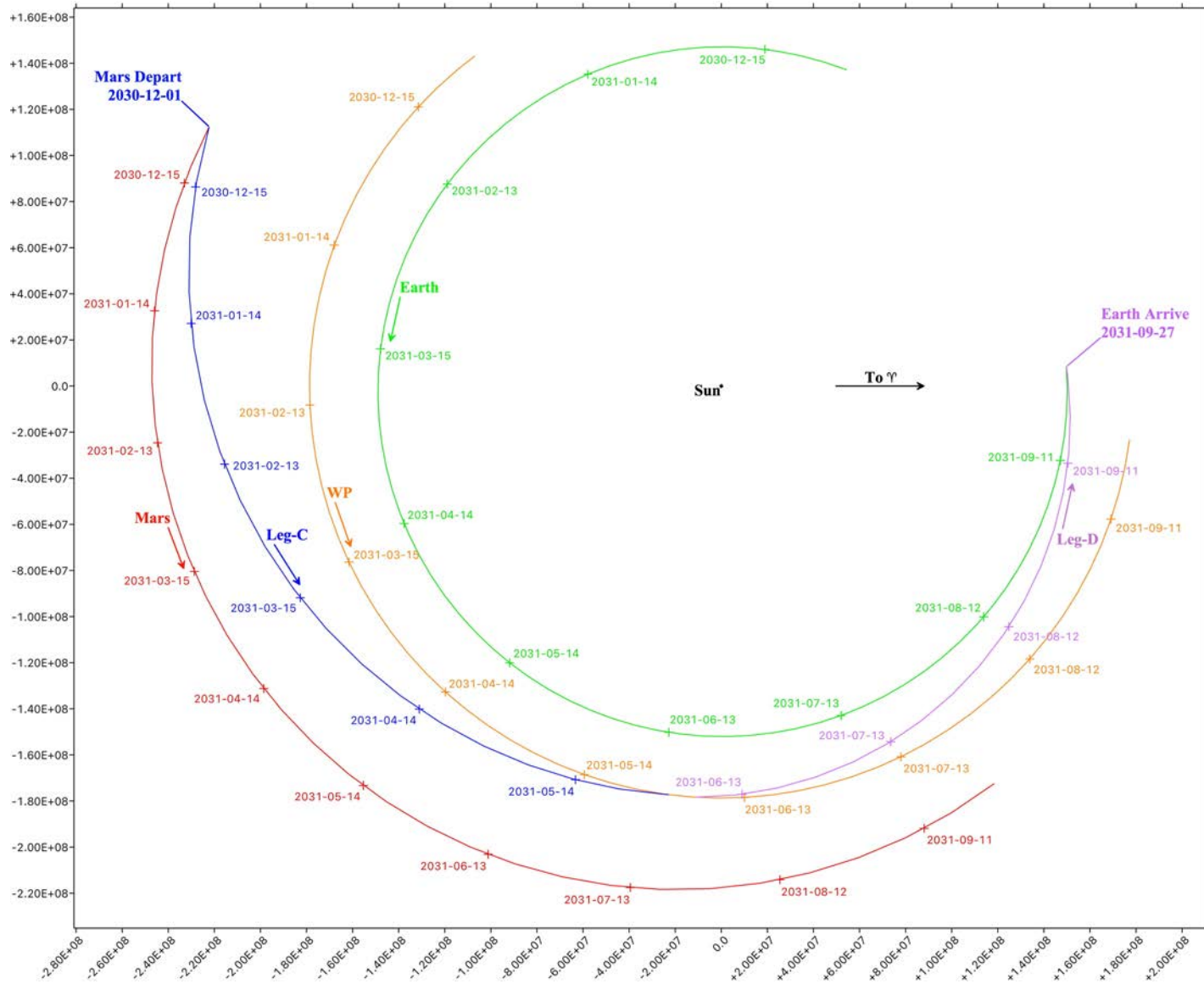
## The $\lambda_{31} = -136.6^\circ$ WP Is Under-Phased For Mars-To-Earth Transits



- Plot is of minimum  $\sum v_\infty$  Leg-CD transit circa 2031 for this WP
- Leg-D must cruise inside the orbit of Earth for more than 90 days because Leg-C had to leave Mars too late

# Interplanetary Synodic-Resonant Waypoints (WPs)

## The $\lambda_{31} = -116.6^\circ$ WP Produces Most Efficient Mars-To-Earth Transits



- Plot is of minimum  $\sum v_\infty$  Leg-CD transit circa 2031 for this WP
- Note this transit always remains between the orbits of Earth and Mars

## Interplanetary Synodic-Resonant Waypoints (WPs)

### Two Optimally Phased $\lambda_{33}$ WPs Support Interleaved Transits Starting Circa 2033

- Therefore, a constellation of 4 WPs supports all Leg-AB and Leg-CD transits

WP Phasing	Supported Transits	Supported Opposition Seasons
$\lambda_{31} = -156.6^\circ$	Leg-AB	2031, 2035, 2040, 2044, ...
$\lambda_{31} = -116.6^\circ$	Leg-CD	2031, 2035, 2040, 2044, ...
$\lambda_{33} = -93.8^\circ$	Leg-AB	2033, 2037, 2042, 2046, ...
$\lambda_{33} = -63.8^\circ$	Leg-CD	2033, 2037, 2042, 2046, ...

- Note variations in  $\sum v_\infty$  due to Mars heliocentric distance at transit arrive (Leg-B)/depart (Leg-C)<sup>4</sup>

Opposition Season	Leg-AB WP			Leg-CD WP		
	Phasing	Minimum $\sum v_\infty$ (km/s)		Phasing	Minimum $\sum v_\infty$ (km/s)	
2031	$\lambda_{31} = -156.6^\circ$	9.480	<i>6.747</i>	$\lambda_{31} = -116.6^\circ$	12.279	<i>7.226</i>
2033	$\lambda_{33} = -93.8^\circ$	7.729	<i>6.336</i>	$\lambda_{33} = -63.8^\circ$	11.310	<i>6.033</i>
2035	$\lambda_{31} = -156.6^\circ$	13.797	<i>5.855</i>	$\lambda_{31} = -116.6^\circ$	8.774	<i>5.980</i>
2037	$\lambda_{33} = -93.8^\circ$	11.318	<i>7.013</i>	$\lambda_{33} = -63.8^\circ$	8.945	<i>6.930</i>
2040	$\lambda_{31} = -156.6^\circ$	15.044	<i>7.206</i>	$\lambda_{31} = -116.6^\circ$	12.224	<i>5.950</i>
2042	$\lambda_{33} = -93.8^\circ$	13.985	<i>5.772</i>	$\lambda_{33} = -63.8^\circ$	13.264	<i>5.654</i>
2044	$\lambda_{31} = -156.6^\circ$	12.545	<i>5.811</i>	$\lambda_{31} = -116.6^\circ$	12.857	<i>7.122</i>
2046	$\lambda_{33} = -93.8^\circ$	12.545	<i>6.416</i>	$\lambda_{33} = -63.8^\circ$	12.441	<i>7.507</i>

<sup>4</sup> Italicized  $\sum v_\infty$  values pertain to contemporaneous single-leg (nonstop) transits to which WP phasing  $\lambda_{31}$  and  $\lambda_{33}$  values do not apply.

# Interplanetary Synodic-Resonant Waypoints (WPs)

## Variations In Maximum Deliverable Payload Mass Are Complex Between Nonstop And Contemporaneous 2-Leg Itineraries Over $T_S$

- Assume propellant mass  $m_P$  can be replenished and payload mass  $m_Y$  can be exchanged/augmented during a 5-day WP loiter
- Assume transits flown by vehicle with inert mass  $m_i = 100,000$  kg, maximum total mass  $m_X = 1,300,000$  kg, and specific impulse  $I_{SP} = 380$  s (producing propulsive exhaust speed  $v_{EX} = 3.726546$  km/s)<sup>5</sup>
- Assume propellant mass  $m_P$  and  $m_Y$  can be freely exchanged such that  $m_i + m_P + m_Y = m_X$  at a departure and  $m_P = 0$  at an arrival
- Assume Earth departures/arrivals from a circular parking orbit at height of +185 km and Mars departures/arrivals from a circular parking orbit at height +382 km
- Document mission parameters for the nonstop and 2-leg minimum  $\sum v_\infty$  transits originating at Earth and Mars during each opposition season from 2031 through 2046
  - Transit phase elapsed time (PET) since original departure at the end of each leg in the transit
  - Maximum deliverable  $m_Y$  at the end of each leg in the transit

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<sup>5</sup> Reference [https://en.wikipedia.org/wiki/SpaceX\\_Starship](https://en.wikipedia.org/wiki/SpaceX_Starship) (accessed 9 Nov 2024).

## Interplanetary Synodic-Resonant Waypoints (WPs)

### Variations In Maximum Deliverable Payload Mass Are Complex Between Nonstop And Contemporaneous 2-Leg Itineraries Over $T_S$ (continued)

- Earth-to-Mars minimum  $\sum v_\infty$  transits

Opposition Year	Itinerary	Waypoint Arrival		Mars Arrival	
		PET (d)	$m_Y$ (kg)	PET (d)	$m_Y$ (kg)
2031	Nonstop			285	142,566
	Leg-AB	145	218,034	300	<b>295,439</b>
2033	Nonstop			200	154,848
	Leg-AB	145	198,011	300	<b>351,466</b>
2035	Nonstop			200	176,938
	Leg-AB	120	195,250	300	<b>63,817</b>
2037	Nonstop			220	149,804
	Leg-AB	125	194,784	300	<b>155,138</b>
2040	Nonstop			300	143,770
	Leg-AB	120	179,231	300	<b>69,288</b>
2042	Nonstop			300	179,238
	Leg-AB	115	101,454	300	<b>84,509</b>
2044	Nonstop			300	174,951
	Leg-AB	125	176,186	300	<b>153,424</b>
2046	Nonstop			290	154,442
	Leg-AB	130	118,788	300	<b>109,614</b>

## Interplanetary Synodic-Resonant Waypoints (WPs)

### Variations In Maximum Deliverable Payload Mass Are Complex Between Nonstop And Contemporaneous 2-Leg Itineraries Over $T_S$ (continued)

- Mars-to-Earth minimum  $\sum v_\infty$  transits

Opposition Year	Itinerary	Waypoint Arrival		Earth Arrival	
		PET (d)	$m_Y$ (kg)	PET (d)	$m_Y$ (kg)
2031	Nonstop			235	143,558
	Leg-CD	180	134,638	300	<b>181,906</b>
2033	Nonstop			215	174,623
	Leg- CD	160	157,282	300	<b>211,818</b>
2035	Nonstop			195	168,506
	Leg- CD	145	281,879	300	<b>217,104</b>
2037	Nonstop			270	138,276
	Leg- CD	145	286,771	300	<b>174,225</b>
2040	Nonstop			285	166,496
	Leg- CD	170	146,322	300	<b>113,668</b>
2042	Nonstop			300	178,882
	Leg- CD	180	96,468	300	<b>112,222</b>
2044	Nonstop			300	142,240
	Leg- CD	185	108,465	300	<b>140,910</b>
2046	Nonstop			245	136,992
	Leg- CD	180	119,756	300	<b>175,799</b>

## Interplanetary Synodic-Resonant Waypoints (WPs)

**Variations In Maximum Deliverable Payload Mass Are Complex Between Nonstop And Contemporaneous 2-Leg Itineraries Over  $T_S$  (concluded)**

Itinerary	Total $m_Y$ Deliverable Over $T_S$ (15 years)	
	To WP (kg)	To Mars (kg)
Nonstop		1,276,558
Leg-AB	1,381,738	<b>1,282,696</b>

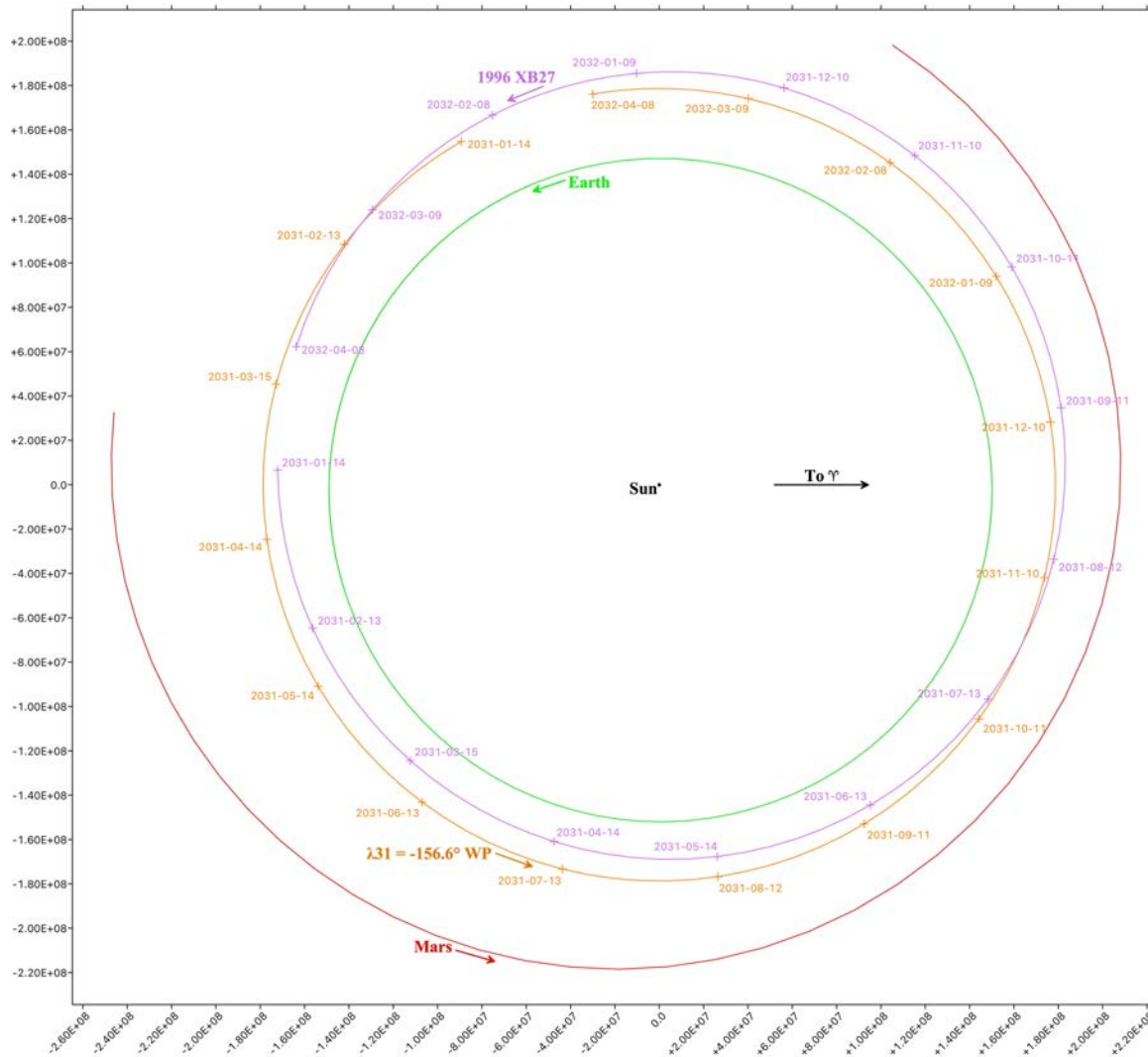
Itinerary	Total $m_Y$ Deliverable Over $T_S$ (15 years)	
	To WP (kg)	To Earth (kg)
Nonstop		1,249,572
Leg-CD	1,331,582	<b>1,327,652</b>

- Are similar economics in play with U.S. commercial air travel? Only about 4% of bookable East/West Coast passenger itineraries are nonstop among the 4 most-patronized airlines.



# Interplanetary Synodic-Resonant Waypoints (WPs)

## Known Asteroids To Build & Supply WPs Are Abundant & Accessible<sup>6</sup>

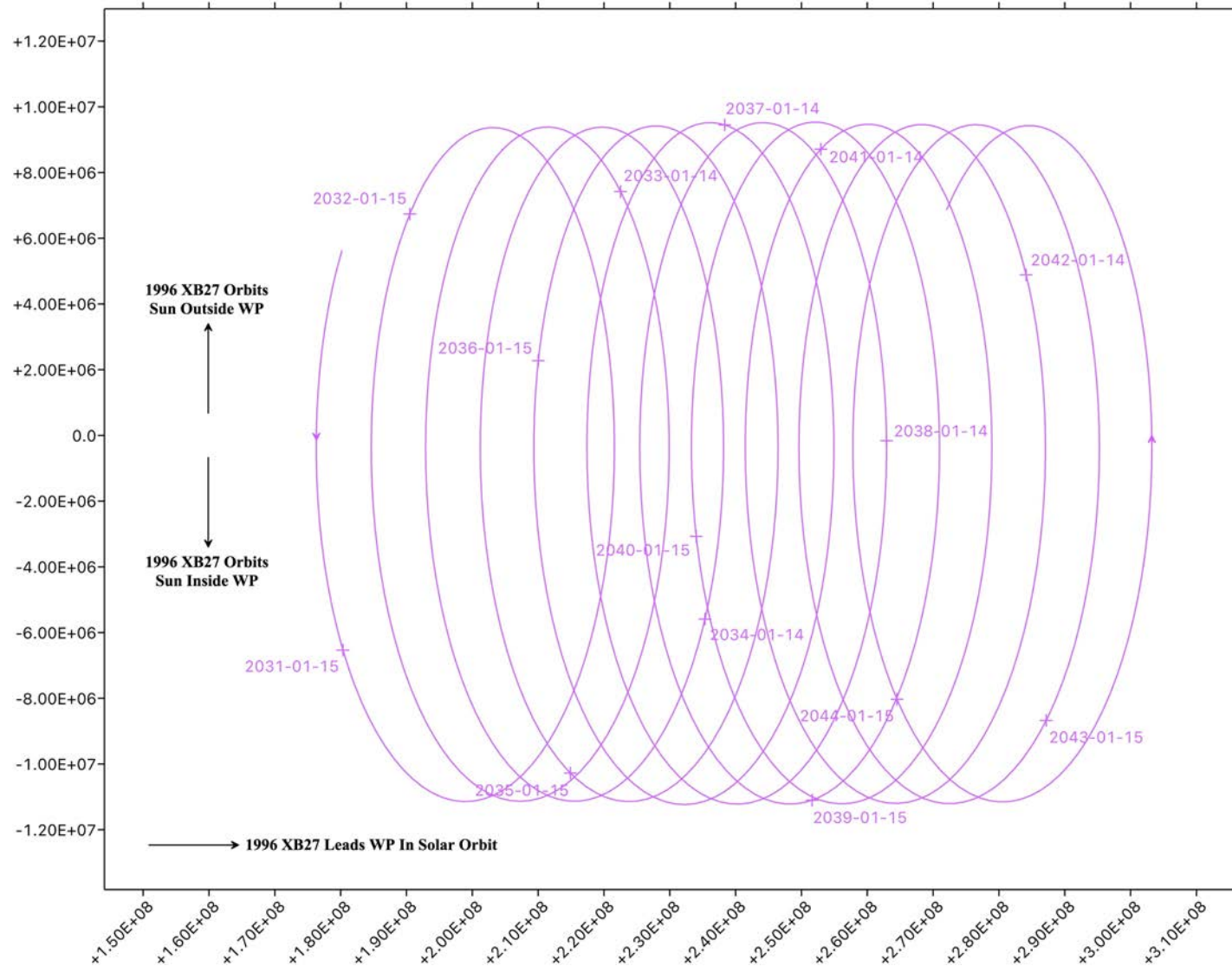


- The rate of asteroid discoveries will only increase as new survey telescopes become operational
- In this example, the  $\lambda_{31} = -156.6^\circ$  WP (orange) is to be supplied from asteroid 1996 XB<sub>27</sub> (violet)
- The estimated diameter of 1996 XB<sub>27</sub> is 100 to 245 meters

<sup>6</sup> As of 12 March 2025, catalogued NEAs orbiting the Sun between Earth and Mars numbered 493.

# Interplanetary Synodic-Resonant Waypoints (WPs)

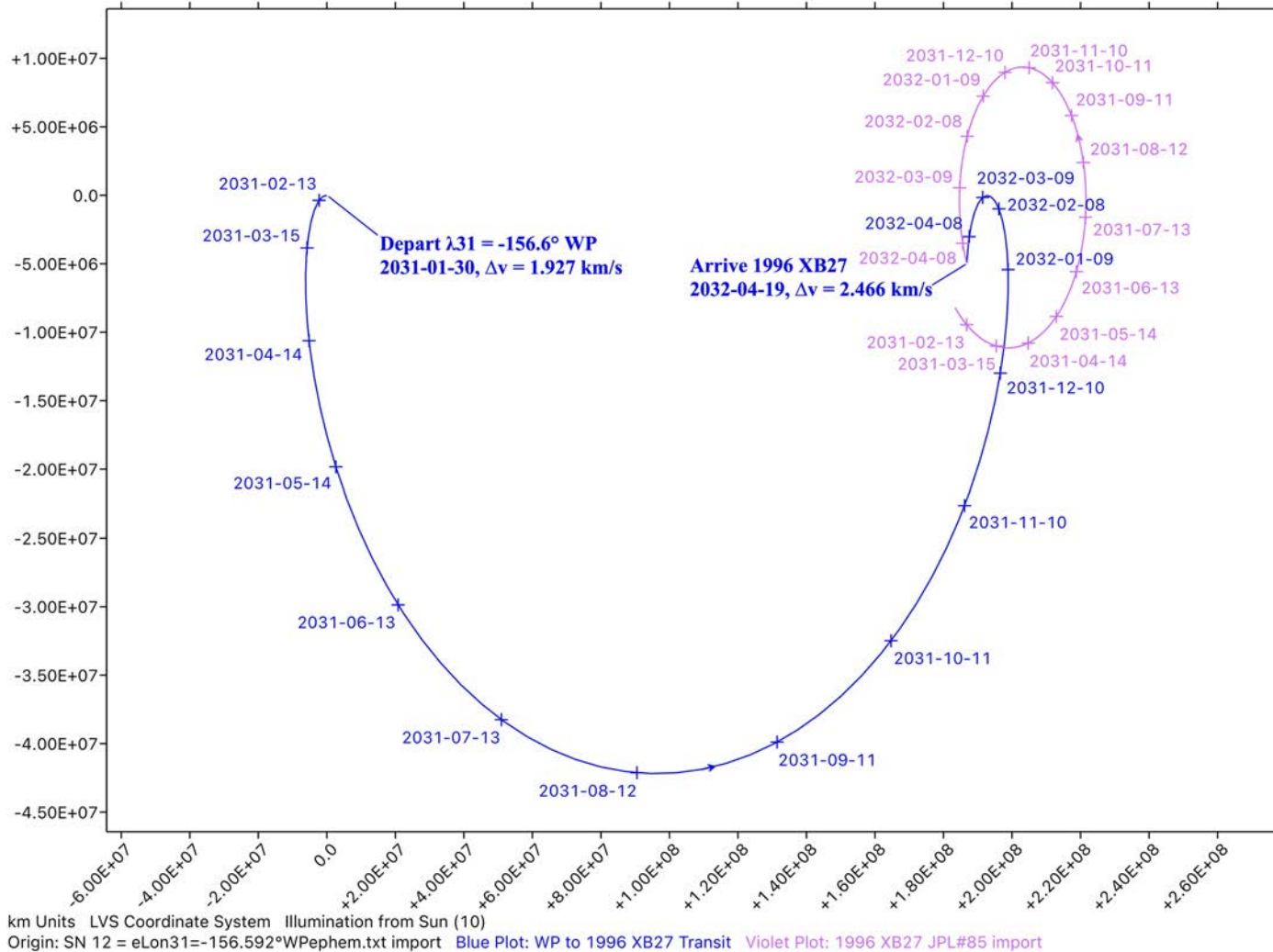
## Known Asteroids To Build & Supply WPs Are Abundant & Accessible (continued)



- In 2031, 1996 XB<sub>27</sub> leads the WP in orbit about the Sun by more than 170 million km
- The lead increases by < 7 million km per Earth year

# Interplanetary Synodic-Resonant Waypoints (WPs)

## Known Asteroids To Build & Supply WPs Are Abundant & Accessible (continued)

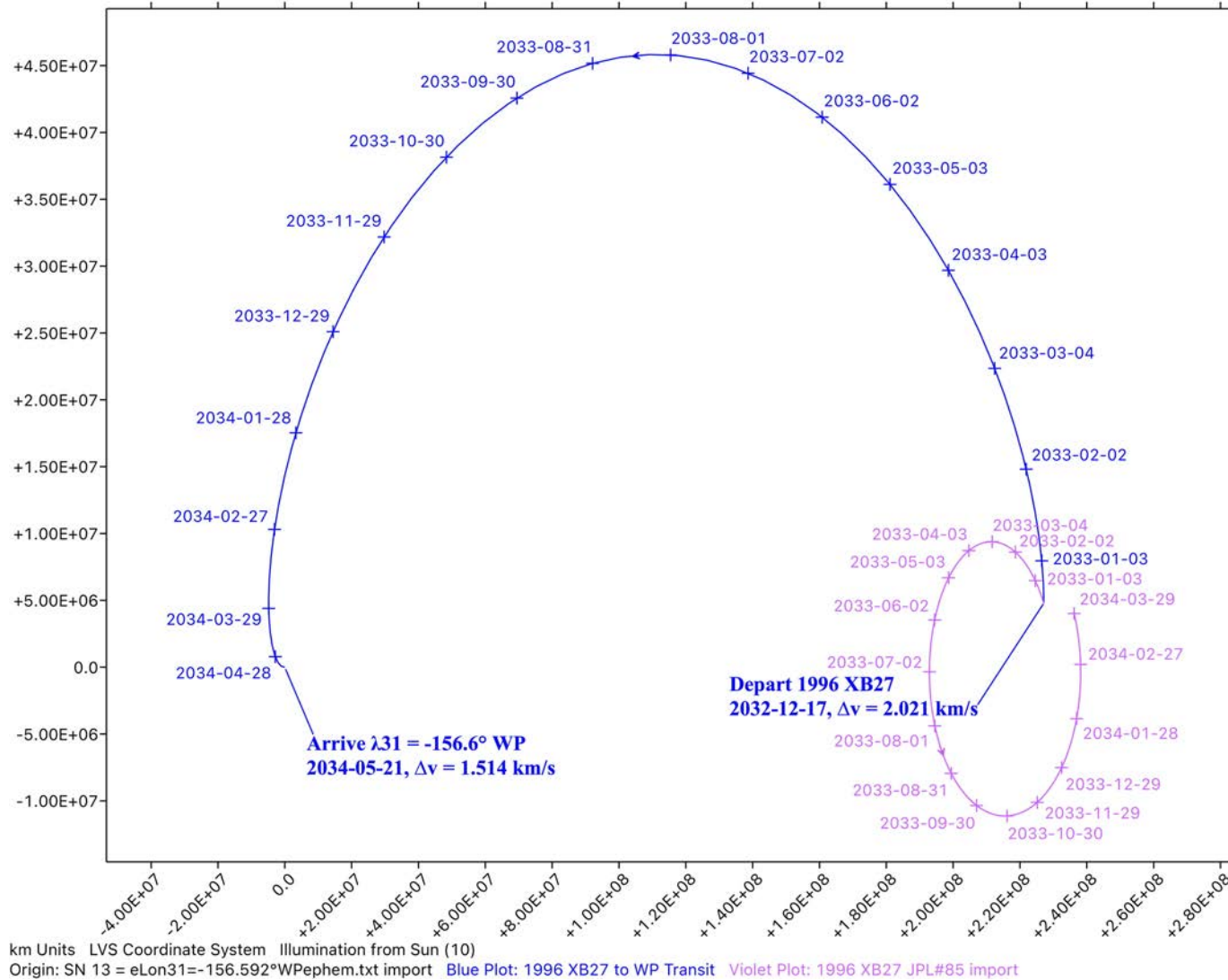


- Outbound transit to 1996 XB<sub>27</sub> requires 445 days
- One-way  $m_Y = 299,926 \text{ kg}^7$
- Leads to 242-day loiter at 1996 XB<sub>27</sub>

<sup>7</sup> For reference, International Space Station mass is near 400,000 kg

# Interplanetary Synodic-Resonant Waypoints (WPs)

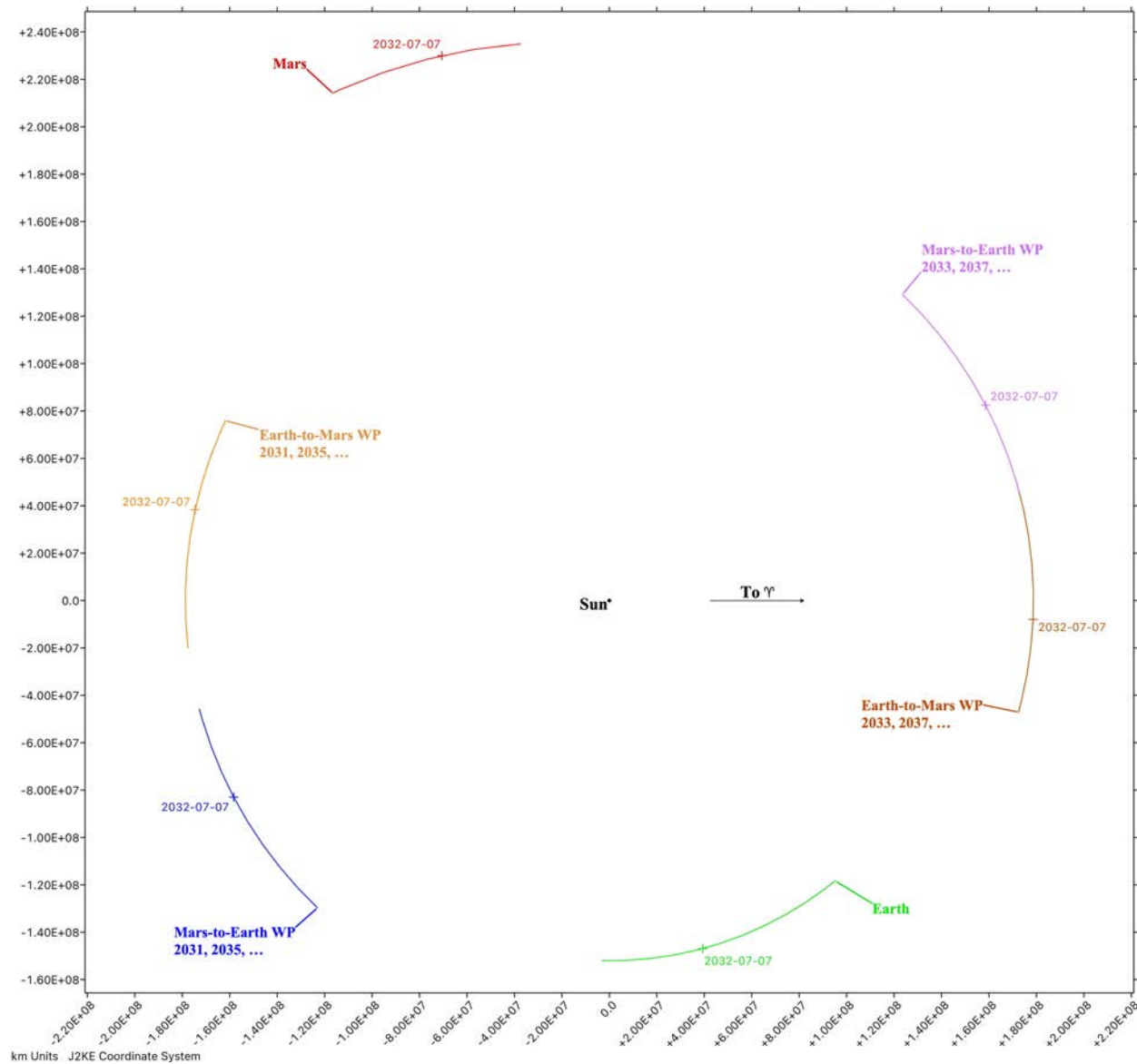
## Known Asteroids To Build & Supply WPs Are Abundant & Accessible (concluded)



- Return transit to WP requires 520 days
- One-way  $m_Y$  = 403,468 kg
- Roundtrip duration = 1207 days = 3.3 years

# Interplanetary Synodic-Resonant Waypoints (WPs)

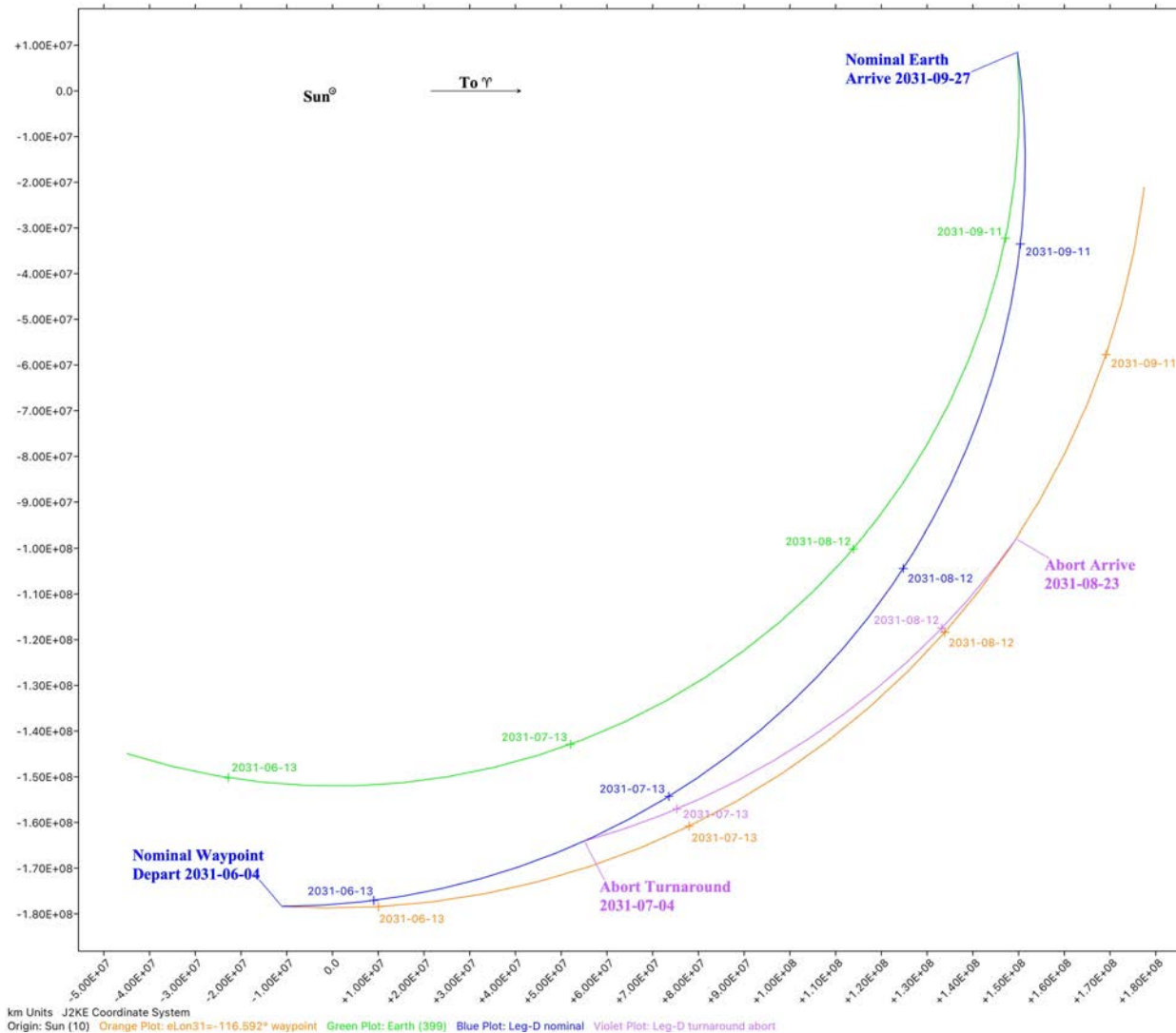
## WPs Serve As Communications Relays During Earth/Mars Solar Conjunctions



- Line-of-sight between Earth and Mars is too close to the Sun for communications from 39 to 51 days during every  $\tau$
- At these times, any of the four WPs can serve as a "bent pipe" communication relay

# Interplanetary Synodic-Resonant Waypoints (WPs)

## WPs Offer Proximal Haven Following Interplanetary Transit Abort



- This scenario commences with 2031's minimum  $\sum v_\infty$  Leg-D nominal trajectory
- Thirty days after departing the  $\lambda_{31} = -116.6^\circ$  WP for Earth, a return abort is initiated with jettison to  $m_Y = 0$
- WP return abort is achieved 80 days after nominal WP departure with +27,323 kg propellant margin

# **Interplanetary Synodic-Resonant Waypoints (WPs)**

## **WPs Support Interplanetary Exploration And Commerce In Many Ways**

- A WP loiter breaks any nonstop transport consumption requirement into two smaller staged segments
- Four WPs serve every Earth/Mars transit opportunity. Each WP is utilized every 4.27 years (on average) without modifications to the WP orbit.
- More cargo mass can be transported to Earth, Mars, and WPs over time than with nonstop Earth/Mars transits using the same spacecraft
- Mass for WP construction and resupply can be obtained from adjacent near-Earth asteroids
- Any WP can serve as an Earth/Mars communications relay to eliminate outages that would otherwise arise when the two planets are on opposite sides of the Sun
- A nominally utilized WP also provides additional forward and turnaround abort destination options

**Original Waypoints Research Published In AIAA's *Journal of Spacecraft and Rockets* (ref. "Synodic-Resonant Earth/Mars Waypoints" At <https://doi.org/10.2514/1.A36186>)**